

Numerical schemes for the simulation of Inertial Confinement Fusion in the direct drive context

Pierre-Henri Maire, maire@celia.u-bordeaux1.fr

UMR CELIA, CNRS, University Bordeaux 1, CEA, 351, cours de la Libération, 33405 Talence, France Collaborations: Rémi Abgrall, Jérôme Breil, Stéphane Galera, Guy Schurtz



- Context of Inertial Confinement Fusion
- Physical models
- Hydrodynamics code
- A cell-centered Lagrangian scheme
- Arbitrary Lagrangian Eulerian scheme
- Conclusions and perspectives

CELIA Laboratory (Bordeaux)



- In 2003, new French policy for lasers and plasmas
- Open CEA's facilities LIL and LMJ to academic users both national and European
 - Demonstrate fusion ignition in laboratory
 - Do basic science research: high energy density physics, astrophysics, hydrodynamics, plasma physics
- Create CELIA laboratory at University Bordeaux I
 - Parent bodies: CEA, CNRS and University Bordeaux I
 - 60 researchers, professors, docs and post-docs
 - Main topics: theoretical and experimental studies in the domain of laser plasma interaction and Inertial Confinement Fusion (Direct Drive)
- Need for CELIA laboratory of a 2D ICF code to prepare and reproduce the experiments on laser created plasmas



 Controlled fusion to produce energy [Lindl, Inertial Confinement Fusion, Springer (1998)]

 $D + T \longrightarrow {}^{4}He$ (3.5MeV) + n (14.1MeV)

Two conditions to obtain fusion

Temperature $T_i > T_{ign}$, typical value $T_i \approx 60 \ 10^6 \text{ K}$ Confinement $N\tau > 10^{14} \text{ cm}^{-3} \text{s}$, Lawson criterion.

Two approaches to fusion applicable to energy production

Magnetic Confinement Fusion $\tau \approx 1 \,\mathrm{s}, N \approx 10^{14} \,\mathrm{cm}^{-3}$. Inertial Confinement Fusion $\tau \approx 10^{-11} \,\mathrm{s}, N \approx 10^{25} \,\mathrm{cm}^{-3}$.

 Main facilities: International Thermonuclear Experimental Reactor, Laser Mega Joule, National Ignition Facility



View of the LMJ and LIL facilities (CEA CESTA near Bordeaux)





View of the LMJ laser bays





Picture of a target









Direct Drive

Indirect Drive

Direct drive has a higher laser-target coupling efficiency but is less uniform in laser irradiation due to discrete beams of lasers.

The indirect drive, by soft X rays which are generated at the inner surface of the holraum, can have a higher uniform irradiation to reduce the growth of perturbations due to Rayleigh-Taylor (RT) instabilities.

Phenomenology of Direct Drive ICF





Laser heating

DT compression

Hot spot ignition

Thermonuclear burn

Main issues for Direct Drive ICF





Target design for Direct Drive ICF



HiPER Project [www.hiper-laser.org]

 DT_{cryo} $\rho = 0.25 \text{ g cm}^{-1}$ 0.1 DT_{gas} $\rho = 10^{-4} \, \mathrm{g \, cm^{-3}}$ 0.08 Radius (cm) 90°0 LASER $\lambda_{\text{las}} = 0.351 \,\mu\,\text{m}$ 0.04 $E_{\rm las} = 200 \text{ kJ}$ $E_{\rm abs} = 130 \text{ kJ}$ 0.02 $R_{e} = 1044 \ \mu m$ 0⊾ 0 2e-09 4e-09 6e-09 $R_i = 833 \ \mu m$ 8e-09 1e-08 1.2e-08 Time (s) All DT target R-t diagram

Target design for Direct Drive ICF



HiPER Project



Density profile at stagnation

Pulse shaping

Compression rates $\sim 10^5$ in $\mathrm{DT}_{\mathrm{gas}}$ and $\sim 10^3$ in $\mathrm{DT}_{\mathrm{cryo}}$. Pulse shaping in order to obtain the highest compression rate by minimizing entropy deposition.

Motivations for a Lagrangian formulation

- ICF plasma is created by laser interaction with target
- ICF target is the assembly of multimaterial layers with high aspect ratios
- Multimaterial flows with large displacements, strong shocks and rarefaction waves
- Simulations with large changes of computational domain volume and shape (animation)

 Lagrangian formulation is well adapted to ICF flows Mesh moves with the fluid Shock resolution is increased No mass flux between cell Free surfaces are naturaly treated Interfaces are sharply resolved

- However for too large deformations (shear and vorticity) it appears a lack of robustness (tangled mesh)
- It can be treated by using an Arbitrary Lagrangian Eulerian (ALE) strategy

Physical model



Plasma fluid model in Lagrangian formalism

$$\begin{split} \rho \frac{d\tau}{dt} &- \nabla \cdot \boldsymbol{V} = 0, \\ \rho \frac{d\boldsymbol{V}}{dt} + \boldsymbol{\nabla}(P_{\mathrm{i}} + P_{\mathrm{e}}) = \boldsymbol{0}, \\ \rho \left(\frac{d\varepsilon_{\mathrm{e}}}{dt} + P_{\mathrm{e}}\frac{d\tau}{dt}\right) - \nabla \cdot (\lambda_{\mathrm{e}}\boldsymbol{\nabla}T_{\mathrm{e}}) = \Omega_{\mathrm{ei}}(T_{\mathrm{i}} - T_{\mathrm{e}}) + W_{\mathrm{las}} + W_{\mathrm{rad}} + W_{\mathrm{fus}}^{\mathrm{e}} + \nabla \cdot \boldsymbol{Q}_{\mathrm{e}}^{\mathrm{nl}}, \\ \rho \left(\frac{d\varepsilon_{\mathrm{i}}}{dt} + P_{\mathrm{i}}\frac{d\tau}{dt}\right) - \nabla \cdot (\lambda_{\mathrm{i}}\boldsymbol{\nabla}T_{\mathrm{i}}) = \Omega_{\mathrm{ei}}(T_{\mathrm{e}} - T_{\mathrm{i}}) + W_{\mathrm{fus}}^{\mathrm{i}}, \\ \frac{\partial \boldsymbol{B}}{\partial t} = \boldsymbol{\nabla} \times \left[\boldsymbol{V} \times \boldsymbol{B} + \frac{c}{eN_{\mathrm{e}}}\boldsymbol{\nabla}(N_{\mathrm{e}}T_{\mathrm{e}}) + \frac{c}{e}\beta \cdot \boldsymbol{\nabla}T_{\mathrm{e}} - \frac{c^{2}}{4\pi}\sigma \cdot \boldsymbol{\nabla} \times \boldsymbol{B} \right]. \end{split}$$

 λ_e tensorial conductivity (Spitzer-Härm and Braginskii), β is the thermoelectric tensor and σ is the tensorial resistivity. Tranverse MHD with magnetic pressure and Joule heating neglected. $Q_{\rm e}^{\rm nl}$ is the non local heat flux.



CHIC : 2D ICF code developped at CELIA



Review on 2D Lagrangian schemes



Staggered schemes

- Velocity at grid nodes
- Internal energy formulation and artificial viscosity
- Wilkins scheme [MCP 3, (1964)]
- Compatible discretization [Caramana and Shashkov JCP 146, (1998)]
- Centered schemes
 - CAVEAT scheme [Dukowicz et al. Report LA-10613-MS, (1986)]
 - 2D conservative extension of Godunov scheme
 - Node velocity computed with a least squares procedure
 - Fluxes not consistent with the nodes motion
 - Després Mazeran scheme [ARMA 178, (2005)]
 - Analysis of 2D gas dynamics in fully Lagrangian form
 - Conservative and entropy consistent 2D scheme
 - Strong dependence to the cell aspect ratio
 - Some problems with boundary conditions



- Motivations
 - Develop a new cell-centered Lagrangian scheme
 - Cell-centered momentum is easier to handle in view of ALE extension
- Main features of our scheme
 - Four pressures on each edge
 - 2D nodal solver
 - Conservative and entropy consistent
 - Recovers the classical Godunov acoustic solver for 1D flows
 - First order accurate
 - Second order extension

R. Abgrall, J. Breil, P.-H. Maire, J. Ovadia. A cell-centered Lagrangian scheme for two-dimensional compressible flow problems. *to appear in SIAM Journal of Scientific Computing* 2007; https://hal.inria.fr/inria-00113542.

Governing equations



Lagrangian hydrodynamics

For a control volume $\Omega(t)$ moving with the fluid velocity $\mathbf{V} = (u, v)^t$

$$\begin{split} & \frac{d}{dt} \int_{\Omega(t)} \rho \, \mathrm{d}\Omega = 0, \quad \text{mass conservation} \\ & \frac{d}{dt} \int_{\Omega(t)} \, \mathrm{d}\Omega - \int_{\partial\Omega(t)} \boldsymbol{V} \cdot \boldsymbol{N} \, \mathrm{d}l = 0, \quad \text{volume conservation} \\ & \frac{d}{dt} \int_{\Omega(t)} \rho \boldsymbol{V} \, \mathrm{d}\Omega + \int_{\partial\Omega(t)} P \boldsymbol{N} \, \mathrm{d}l = \boldsymbol{0}, \quad \text{momentum conservation} \\ & \frac{d}{dt} \int_{\Omega(t)} \rho E \, \mathrm{d}\Omega + \int_{\partial\Omega(t)} P \boldsymbol{V} \cdot \boldsymbol{N} \, \mathrm{d}l = 0, \quad \text{total energy conservation} \end{split}$$

The pressure *P* is given by the equation of state $P \equiv P(\rho, \varepsilon)$ where $\varepsilon = E - \frac{1}{2} \mathbf{V} \cdot \mathbf{V}$. Note that the volume conservation equation is also a geometrical law conservation.



• Discrete evolution equations for $(\tau_c = 1/\rho_c, V_c, E_c)$



Note that the cell area satisfies $A_c = m_c \tau_c$.



Nodal approximation of the geometrical law conservation

Compute the area of the cell using a triangular decomposition



$$A_c = \frac{1}{2} \sum_{n \in \mathcal{N}(c)} \left(\boldsymbol{On} \times \boldsymbol{On}_+ \right) \cdot \boldsymbol{e_Z},$$

time differentiation leads to

$$\frac{d}{dt}A_{c} = \sum_{n \in \mathcal{N}(c)} \left(L_{\underline{n}}^{c} N_{\underline{n}}^{c} + L_{\overline{n}}^{c} N_{\overline{n}}^{c} \right) \cdot V_{n}^{\star},$$

where $\mathcal{N}(c)$ is the set of vertices of cell c.

We need to compute first the nodal velocity in order to satisfy the geometrical law conservation. This is the only way to ensure the compatibility between vertex motion and cell area variation.



Nodal approximation of the momentum flux

 $P_{\underline{n}}^{\star,c}$ and $P_{\overline{n}}^{\star,c}$ pressures on both sides of node *n*, viewed from cell *c*.



The momentum equation is written

$$m_{c}\frac{d}{dt}\boldsymbol{V_{c}} + \sum_{n \in \mathcal{N}(c)} \left(L_{\underline{n}}^{c} P_{\underline{n}}^{\star,c} \boldsymbol{N_{\underline{n}}^{c}} + L_{\overline{n}}^{c} P_{\overline{n}}^{\star,c} \boldsymbol{N_{\overline{n}}^{c}} \right) = \boldsymbol{0}.$$

 n^{\neg}



• Evolution equations for the discrete unknowns (τ_c, V_c, E_c)

$$m_{c}\frac{d}{dt}\tau_{c} - \sum_{n\in\mathcal{N}(c)} \left(L_{\underline{n}}^{c}\boldsymbol{N}_{\underline{n}}^{c} + L_{\overline{n}}^{c}\boldsymbol{N}_{\overline{n}}^{c} \right) \cdot \boldsymbol{V}_{n}^{\star} = 0,$$

$$m_{c}\frac{d}{dt}\boldsymbol{V}_{c} + \sum_{n\in\mathcal{N}(c)} \left(L_{\underline{n}}^{c}P_{\underline{n}}^{\star,c}\boldsymbol{N}_{\underline{n}}^{c} + L_{\overline{n}}^{c}P_{\overline{n}}^{\star,c}\boldsymbol{N}_{\overline{n}}^{c} \right) = \mathbf{0},$$

$$m_{c}\frac{d}{dt}E_{c} + \sum_{n\in\mathcal{N}(c)} \left(L_{\underline{n}}^{c}P_{\underline{n}}^{\star,c}\boldsymbol{N}_{\underline{n}}^{c} + L_{\overline{n}}^{c}P_{\overline{n}}^{\star,c}\boldsymbol{N}_{\overline{n}}^{c} \right) \cdot \boldsymbol{V}_{n}^{\star} = 0.$$

• Node motion $\frac{d}{dt} X_n = V_n^\star, \ X_n(0) = x_n.$

Construction of a nodal solver to evaluate the nodal fluxes V_n^{\star} , $P_{\underline{n}}^{\star,c}$ and $P_{\overline{n}}^{\star,c}$.





Generally, $P_{\overline{n}}^{\star,c} \neq P_{\underline{n}}^{\star,k}$ and $P_{\underline{m}}^{\star,c} \neq P_{\overline{m}}^{\star,k}$.

Contrarily to 1D Riemann solver approach momentum conservation is not ensured automatically! But we shall show how to recover it.



Remark2 : local entropy inequality

Starting from Gibbs relation $TdS = d\varepsilon + Pd\tau$, we get

$$m_c \left(\frac{d\varepsilon_c}{dt} + P_c \frac{d\tau_c}{dt}\right) = \sum_{n \in \mathcal{N}(c)} \left[L_{\underline{n}}^c (P_c - P_{\underline{n}}^{\star,c}) \mathbf{N}_{\underline{n}}^c + L_{\overline{n}}^c (P_c - P_{\overline{n}}^{\star,c}) \mathbf{N}_{\overline{n}}^c \right] \cdot (\mathbf{V}_n^{\star} - \mathbf{V}_c)$$

A sufficient condition to have an entropy inequality is to set

$$P_{c} - P_{\underline{n}}^{\star,c} = Z_{c} \left(V_{n}^{\star} - V_{c} \right) \cdot N_{\underline{n}}^{c},$$
$$P_{c} - P_{\overline{n}}^{\star,c} = Z_{c} \left(V_{n}^{\star} - V_{c} \right) \cdot N_{\overline{n}}^{c},$$

where $Z_c > 0$ is the acoustic impedance of the cell c. We recover Riemann invariants along directions N_n^c and $N_{\overline{n}}^c$.

$$m_c T_c \frac{dS_c}{dt} = \sum_{n \in \mathcal{N}(c)} Z_c \left\{ L_{\underline{n}}^c [(V_n^{\star} - V_c) \cdot N_{\underline{n}}^c]^2 + L_{\overline{n}}^c [(V_n^{\star} - V_c) \cdot N_{\overline{n}}^c]^2 \right\} \ge 0.$$

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- Weighted least squares procedure for the vertex velocity
- $V_n^{\star} = (u_n^{\star}, v_n^{\star})^t$ minimizes the functional

$$I(u_{n}^{\star}, v_{n}^{\star}) = \sum_{c^{-}, c} \omega_{c^{-}, c} \left(V_{n}^{\star} \cdot N_{c^{-}, c} - \mathcal{V}_{c^{-}, c}^{\star} \right)^{2}.$$

$$N_{c^{-}, c} = N_{\underline{n}}^{c^{-}} = -N_{\overline{n}}^{c}$$

$$V_{c^{-}, c} \text{ weight on edge } [c^{-}, c]$$

$$V_{c^{-}, c}^{\star} \text{ normal Riemann velocity on } [c^{-}, c]$$

$$V_{c^{-}, c}^{\star} = \left(\frac{Z_{c} V_{c} + Z_{c^{-}} V_{c^{-}}}{Z_{c} + Z_{c^{-}}} \right) \cdot N_{c^{-}, c} - \frac{P_{c} - P_{c^{-}}}{Z_{c} + Z_{c^{-}}}$$

$$V_{n}^{\star} \left(\frac{Z_{c}, V_{c}, P_{c}}{P_{\underline{n}}^{\star, c}} - \frac{N_{\underline{n}}^{c^{-}}}{N_{\underline{n}}^{c^{-}}} \right)$$

 $\omega_{c^-,c}$ will be computed so that momentum conservation will be recovered.



Weights evaluation

Least squares problem posed by $\nabla I = 0$, this yields the vectorial equation

$$\sum_{c^{-},c} \omega_{c^{-},c} \left(\boldsymbol{V_{n}^{\star}} \cdot \boldsymbol{N_{c^{-},c}} - \boldsymbol{\mathcal{V}_{c^{-},c}^{\star}} \right) \boldsymbol{N_{c^{-},c}} = \boldsymbol{0}.$$
(1)

Riemann invariants

$$P_{c^{-}} - P_{\underline{n}}^{\star,c^{-}} = Z_{c^{-}} \left(V_{n}^{\star} - V_{c^{-}} \right) \cdot N_{c^{-},c},$$
$$P_{c} - P_{\overline{n}}^{\star,c} = -Z_{c} \left(V_{n}^{\star} - V_{c} \right) \cdot N_{c^{-},c}.$$

By substraction of the Riemann invariants

$$P_{\overline{n}}^{\star,c} - P_{\underline{n}}^{\star,c^{-}} = \left(Z_{c} + Z_{c^{-}}\right) \left(\boldsymbol{V}_{\boldsymbol{n}}^{\star} \cdot \boldsymbol{N}_{\boldsymbol{c}^{-},\boldsymbol{c}} - \boldsymbol{\mathcal{V}}_{c^{-},c}^{\star} \right).$$
(2)

Since $V_n^{\star} \cdot N_{c^-,c} \neq \mathcal{V}_{c^-,c}^{\star}$ then $P_{\overline{n}}^{\star,c} \neq P_{\underline{n}}^{\star,c^-}$ (momentum conservation loss).



Weights evaluation

Substituting (2) in (1)

$$\sum_{c^{-},c} \frac{\omega_{c^{-},c}}{Z_{c}+Z_{c^{-}}} \left(P_{\overline{n}}^{\star,c}-P_{\underline{n}}^{\star,c^{-}}\right) N_{c^{-},c} = \mathbf{0}.$$
(3)

It can be interpretated as a momentum balance around node n provided

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Momentum and energy conservation

With these weights one can check that

$$\sum_{c \in \mathcal{C}(n)} \left(L_{\underline{n}}^{c} P_{\underline{n}}^{\star,c} N_{\underline{n}}^{c} + L_{\overline{n}}^{c} P_{\overline{n}}^{\star,c} N_{\overline{n}}^{c} \right) = \mathbf{0},$$
$$\sum_{c \in \mathcal{C}(n)} \left(L_{\underline{n}}^{c} P_{\underline{n}}^{\star,c} N_{\underline{n}}^{c} + L_{\overline{n}}^{c} P_{\overline{n}}^{\star,c} N_{\overline{n}}^{c} \right) \cdot V_{n}^{\star} = \mathbf{0},$$

where C(n) is the set of the cells sharing node n.

We have recovered momentum and total energy conservation.



Summary of nodal fluxes computation

$$\begin{aligned} \boldsymbol{V}_{\boldsymbol{n}}^{\star} &= \left(\sum_{c^{-},c} \omega_{c^{-},c} \boldsymbol{N}_{\boldsymbol{c}^{-},\boldsymbol{c}} \otimes \boldsymbol{N}_{\boldsymbol{c}^{-},\boldsymbol{c}}\right)^{-1} \left(\sum_{c^{-},c} \omega_{c^{-},c} \mathcal{V}_{\boldsymbol{c}^{-},c}^{\star} \boldsymbol{N}_{\boldsymbol{c}^{-},\boldsymbol{c}}\right), \\ P_{c^{-}} &= P_{\underline{n}}^{\star,c^{-}} = Z_{c^{-}} \left(\boldsymbol{V}_{\boldsymbol{n}}^{\star} - \boldsymbol{V}_{\boldsymbol{c}^{-}}\right) \cdot \boldsymbol{N}_{\boldsymbol{c}^{-},\boldsymbol{c}}, \\ P_{c} &= P_{\overline{n}}^{\star,c} = -Z_{c} \left(\boldsymbol{V}_{\boldsymbol{n}}^{\star} - \boldsymbol{V}_{\boldsymbol{c}}\right) \cdot \boldsymbol{N}_{\boldsymbol{c}^{-},\boldsymbol{c}}, \\ \mathcal{V}_{c^{-},c}^{\star} &= \left(\frac{Z_{c}\boldsymbol{V}_{\boldsymbol{c}} + Z_{c^{-}}\boldsymbol{V}_{\boldsymbol{c}^{-}}}{Z_{c} + Z_{c^{-}}}\right) \cdot \boldsymbol{N}_{\boldsymbol{c}^{-},\boldsymbol{c}} - \frac{P_{c} - P_{c^{-}}}{Z_{c} + Z_{c^{-}}}, \\ \omega_{c^{-},c} &= L_{c^{-},c} \left(Z_{c} + Z_{c^{-}}\right). \end{aligned}$$

We can check that the nodal velocity is always defined provided the mesh is non-degenerate (Schwarz inequality).



ID flow with planar symmetry



Cells 1 and 4: state P_1 , Z_1 , $V_1 = u_1 e_X$. Cells 2 and 3: state P_2 , Z_2 , $V_2 = u_2 e_X$.

Simple calculations yields $A = (L_2 + L_4)(Z_1 + Z_2)$, $B = 2(L_1Z_1 + L_3Z_2)$, C = 0, $SM_X = (L_2 + L_4)(P_1 - P_2 + Z_1u_1 + Z_2u_2)$ and $SM_Y = 0$. Therefore, the nodal velocity is given by

$$u_q^{\star} = \frac{P_1 - P_2 + Z_1 u_1 + Z_2 u_2}{Z_1 + Z_2},$$

$$v_q^{\star} = 0.$$

We recover exactly the 1D Godunov acoustic solution!



ID flow with cylindrical symmetry



Consider an equi-angular polar mesh. Choose the local frame (M_q, e_X, e_Y) . Cells 1 and 4: state P_1 , Z_1 , $V_1 = V_1 N_2^1$, $V_4 = -V_1 N_4^3$. Cells 2 and 3: state P_2 , Z_2 , $V_2 = V_2 N_2^1$, $V_3 = -V_2 N_4^3$.

The cylindrical symmetry provides

$$u_q^{\star} = \frac{P_1 - P_2 + Z_1 V_1 + Z_2 V_2}{Z_1 + Z_2} \frac{1}{\cos(\frac{\theta}{2})},$$

$$v_q^{\star} = 0.$$

Once more the Godunov acoustic solver modified by a geometrical factor.



Comments

- Vertex velocity and fluxes computed consistently
- ◊ Local entropy inequality
- ◊ Momentum and total energy conservation
- Naturally unstructured
- Recover classical Godunov acoustic solver for 1D flows
- ◊ Boundary conditions easily derived
- ◊ Tabulated EOS provided the isentropic sound speed is known.
- ◊ Spatial approximation first order accurate



• Piecewise linear reconstruction for P and V using P_c and V_c

$$\phi_c(\boldsymbol{X}) = \phi(\boldsymbol{X}_c) + \boldsymbol{\nabla}\phi_c \cdot (\boldsymbol{X} - \boldsymbol{X}_c),$$

where X_c is defined by $\int_c (X - X_c) dX dY = 0$ and $\phi(X_c) = \phi_c$.

- Calculation of $\nabla \phi_c$ by a least square procedure
- Reconstruction exact for linear fields
- Monotony ensured by slope limiters so that

$$\min_{k \in \mathcal{C}(c)} \phi_k \le \phi_c(\boldsymbol{X_n}) \le \max_{k \in \mathcal{C}(c)} \phi_k,$$

where C(c) is the set of the nearest neighbors of cell c.

- Nodal extrapolated values for the nodal solver
- Explicit time discretization by a 2 steps Runge-Kutta procedure



• Sod shock tube, first and second order calculation for $n_x = 100$ cells





Saltzman test case



Computational domain $(x, y) \in [0, 1] \times [0, 0.1]$ with $(n_x, n_y) = (100, 10)$ stretched by the map

$$x_{str} = x + (0.1 - y)\sin(x\pi),$$

$$y_{str} = y.$$

Initial conditions: $(\rho^0, P^0, V^0) = (1, 0, 0)$, perfect gas with $(\gamma = 5/3)$. Inflow velocity $V^* = -1$ on the left boundary x = 0. Planar strong shock wave with speed $D = \frac{4}{3}e_x$.



Saltzman test case



Mesh at t = 0.6



Mesh at t = 0.75



Saltzman test case







Mesh at t = 0.96





• Noh test case on a 50×50 Cartesian grid



Mesh at t = 0.6



Zoom near the center







Density map (left) and density in all the cells (right) at t = 0.6.



Sedov test case on a 50×50 Cartesian grid



Density map (left) and density in all the cells (right) at t = 1. Ref : R. LOUBÈRE, M.J. SHASHKOV, A subcell remapping method on staggered polygonal grids for ALE methods, JCP 209 (2005) 105-138.



Sedov test case on a polygonal grid



Density map (left) and density in all the cells (right) at t = 1.



Linear phase of Richtmyer-Meshkov instability

Ref : Y. YANG, Q. ZHANG, D. SHARP, *Small amplitude theory of Richtmyer-Meshkov instability*, Phys. Fluids 6(5), May 1994. $P^* = 2$



Perturbation amplitude evaluated with 2D computations

$$a(t) = \left(x_{\text{pert}}(t) - x_{\text{unpert}}(t)\right) / a_0$$

- ◊ First and second order calculations
- ◊ Comparison with linear theory



X-t diagram of the flow





Perturbation amplitude evolution



Comparison between first, second order and linear theory

CELIA

Rayleigh-Taylor and RM instability for a spherical implosion

Ref: D. YOUNGS, *3D Numerical simulation of turbulent mixing in spherical implosions*, 10th IWPCTM, Paris July 2006.



◇ Initial perturbation at internal interface with Legendre modes

$$r_{\rm i}^{\rm pert} = r_{\rm i} \left[1 + a_0 P_l(\cos \theta) \right]$$

♦ We set $a_0 = 10^{-3}$ and l = 10



R-t diagram of the spherical implosion





Development of the RT instability







Final grid



Development of the RT instability



Experiment design



Impact of irradiation symmetry on cylinder compression



 $\lambda_{\text{las}} = 0.53 \,\mu\text{m}$, 1 ns square pulse (animation).

Submitted expirement (Vulcan laser facility, Rutherford Appleton Laboratory, 2008, HIPER project).

Experiment design



Impact of irradiation symmetry on cylinder compression



Experiment design



Impact of irradiation symmetry on cylinder compression



Experiment restitution



- Characterization of non local heat transport
- Several shots on LIL laser (CEA Bordeaux, December 2005)



- ♦ Plastic target with Vanadium and Titanium markers of 200 μm thickness
- ♦ Vanadium marker of $0.1 \ \mu m$ thickness at $5 \ \mu m$
- $\diamond~$ Titanium marker of $0.1~\mu m$ thickness at $15~\mu m$
- Include steady state radiative physics and MHD
- ♦ LIL laser beam : $\lambda = 0.35 \ \mu m$, maximal intensity 1. $10^{15} W \ cm^{-2}$

Ref : G. SCHURTZ ET AL, *Revisiting nonlocal electron-energy transport in Inertial Fusion conditions*, PRL 98, 2007.

Experiment restitution



Map of electronic temperature



Motivations for ALE methodology



- Lagrangian computations can degenarate
- Degenerate typicaly for shear or rotational flows (high velocity impact)
- Can be treated by Arbitrary Lagrangian-Eulerian method







- ALE: combination of Lagrangian and Eulerian methods (Hirt, Amsden, Cook JCP 1974, 1997)
 - Lagrangian computation (several time step)
 - Rezoning : mesh untangling and smoothing
 - Remapping : conservative interpolation of conservative variables from Lagrangian to rezoned mesh
- Remapping allows mass flux between cells
- ALE strategy combines good features of both Lagrangian and Eulerian approaches



- Goal : improvement of unstructured mesh quality
- Minimization of nodally based objective functions (Knupp IJNME 48 2000)
- Local objective function based on geometric entities associated with the node
- Main issue : how can we incorporate physical criteria in rezoning?





Nodally based objective function



Objective function based on *the condition number of the Jacobian matrix*

$$F_{s}(x,y) = \frac{1}{2} \sum_{k} \|\mathbf{J}_{k}^{-1}\| \|\mathbf{J}_{k}\|,$$

where |||| is the Frobenius norm. F_s is the discrete analog of the *smoothness* functional proposed by Brackbill (JCP 46, 1982). It is closely related to the Winslow smoother.

$$F_s(x,y) = \frac{1}{2} \sum_k \frac{L_k^2 + L_{k+1}^2}{|\mathsf{J}_k|}.$$



Nodally based objective function



In the same framework we derive a weighted orthogonality control functional

$$F_o(x,y) = \frac{1}{2} \sum_{k} \frac{(e_k \cdot e_{k+1})^2}{|\mathsf{J}_k|^2}.$$

 F_o is the discrete analog of the *orthogonality control functional* proposed by Brackbill (JCP 46, 1982).



Nodally based objective function



 $m{e}_{m{k}} = m{x}_{m{k}} - m{x}$, edge vector ${\sf J}_{\sf k} = [m{e}_{m{k}}, m{e}_{m{k+1}}]$, Jacobian matrix

We also derive a weighted area control functional

$$F_w(x,y) = \frac{1}{2} \sum_k W_k | \mathbf{J}_k |^2,$$

W is a weight function resulting in a finer grid where the weight is large.

Contrarily to F_s and F_o , F_w has a physical dimension. Consequently, it is difficult to combine these 3 objective functions.



- Optimization of the global mesh
 - \diamond Construct a local objective function F(x,y) by combining $F_s,\,F_o$ and F_w
 - Minimize separately each of the local objective function with a Newton procedure (single step towards the minimum)
 - ◊ Iterate over all the nodes of the mesh
- Treatment of boundary and interface nodes
 - ◊ Boundaries and interfaces are fitted with Bezier curves
 - ◊ Nodes are assigned to move on these Bezier curves
 - ◊ It leads to constrained minimization problem
- Mesh untangling by combination of feasible set method and numerical optimization [Vachal, Garimella, Shashkov (JCP, 2004)].
- Main issue : we need to move only those vertices which are necessary and as little as possible

Remapping



- Conservative interpolation of conservative quantities from the old Lagrangian mesh to the new rezoned mesh
- Piecewise monotonic linear reconstruction over the Lagrangian mesh
- Approximate quadrature over regions swept by edges moving from Lagrangian position to the rezoned position
- Integral over new cells=sum of integrals over swept regions
- Cheaper than exact integration for wich intersections is needed





• Sedov test case on a 50×50 Cartesian grid







Euler

Lagrange





• Sedov test case on a 50×50 Cartesian grid





Non-linear phase of Richtmyer-Meshkov instability

Ref : C. MÜGLER, L. HALLO ET AL., Validation of an ALE Godunov Algorithm for Solutions of the Two-Species Navier-Stokes Equations, AIAA 96-2068, June 1996.



- ♦ Initial perturbation $X_{inter} = 0.4 + a_0 \cos(\frac{2\pi}{\lambda}Y)$
- ◇ ALE second order calculation
- ◊ Mixture assumption isoP, isoT
- $\diamond~$ 3 Meshes, cell sizes: $2\times2~mm^2$, $1\times1~mm^2$ and $0.5\times0.5~mm^2$



X-t diagram of the flow





Non-linear phase of Richtmyer-Meshkov instability





High velocity impact test case





- An accurate and robust 2D cell-centered second order Lagrangian scheme
- Axisymmetric extension has been achieved
- Coupling with diffusion scheme
- ALE method
- Future works
 - Improvement of rezoning procedure (including physical criteria)
 - Improvement of the nodal solver
 - Three dimensional Lagrangian scheme
 - Interface reconstruction (VOF method)
 - Mixed cells treatment